

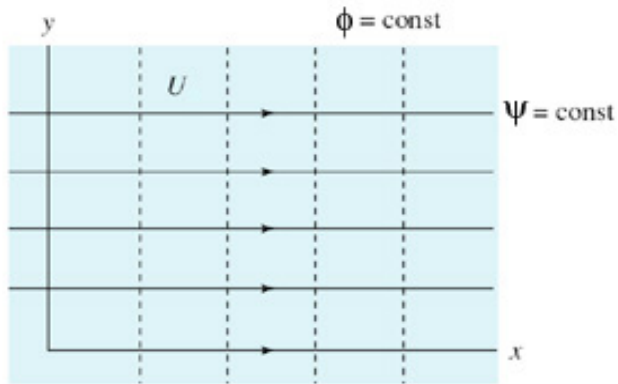
# Superposition

Section 8.5.3

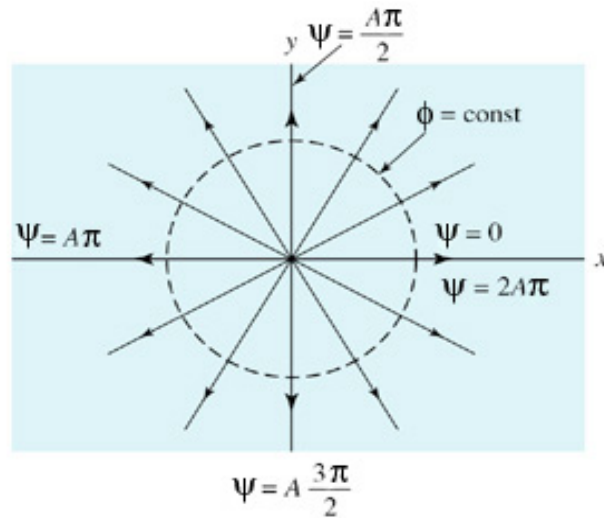
# Simple Potential Flows

- Most complex potential (inviscid, irrotational) flows can be modeled using a combination of simple potential flows
- The simple flows used are:
  - Uniform flows
  - Line sources and sinks
  - Irrotational vortices
  - Doublets
- All of these simple flows satisfy the Laplace equation

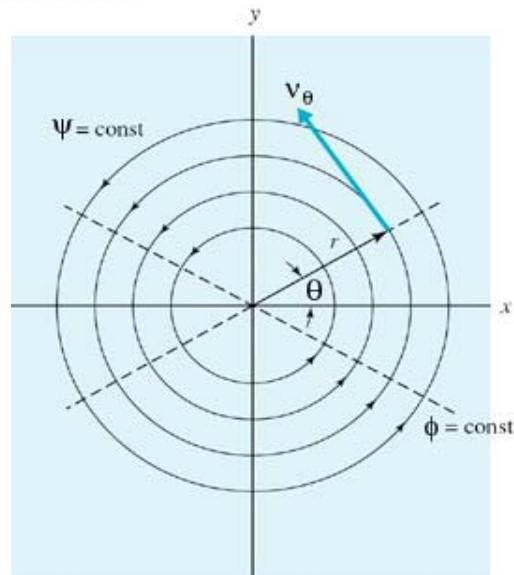
# Simple Potential Flows



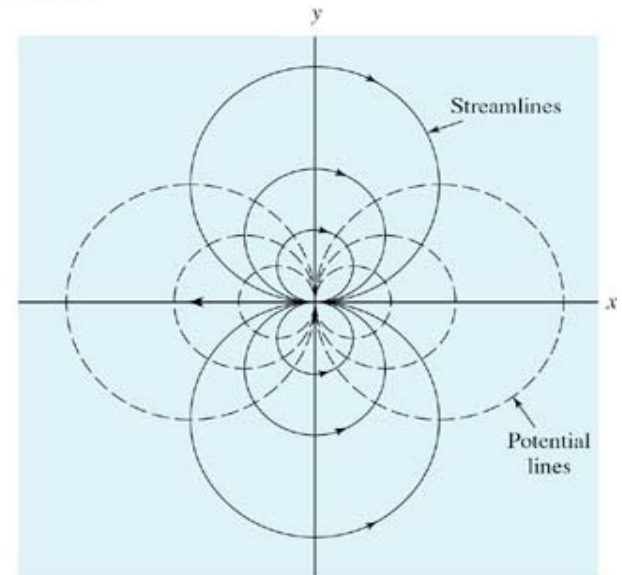
(a) Uniform flow in  $x$ -direction



(b) Line source



(c) Irrotational vortex



(d) Doublet

# Superposition of potential flows

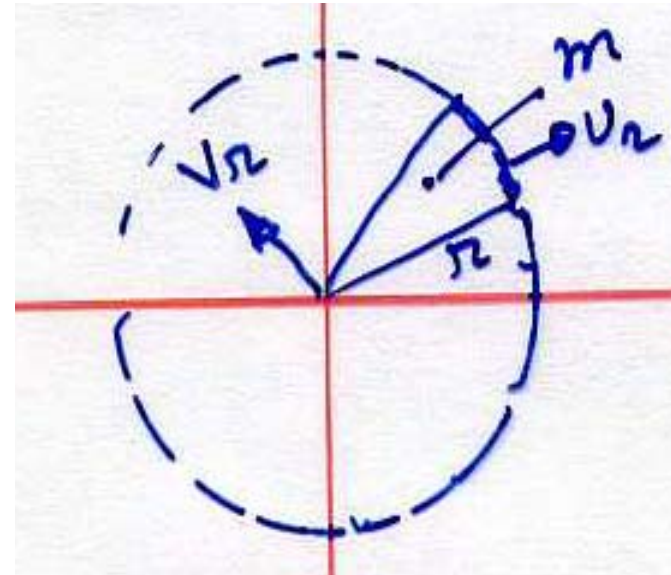
- If  $\psi_1$  and  $\psi_2$  are two independent solutions of Laplace eqn then  $\psi = \psi_1 + \psi_2$  satisfies Laplace eqn.
- So stream function of a complex flow  $\psi$  can be found by adding stream functions ( $\psi_1, \psi_2$ ) of two simpler flows

Note: The above conclusion is also valid for potential functions

# Sources & Sinks

- If fluid flows radially outward it is a **source**
  - For radially inward it is **sink**.
  - Let  $q$  be the volume flow rate per unit length of a line passing through center
- As there is no flow in angular direction

$$v_{\theta} = 0 = -\frac{\partial \psi}{\partial r} = \frac{1}{r} \frac{\partial \phi}{\partial \theta}$$



Therefore the volume flow (per unit length) of a source/sink is only in the radial direction:

$$2 * \pi * r * v_r = q \quad [\text{Note } q \text{ is -ve for Sink}] \quad \text{Here } q \text{ is called Strength}$$

$$v_r = \frac{q}{2\pi r} = \frac{1}{r} \frac{\partial \psi}{\partial \theta} = \frac{\partial \phi}{\partial r}$$

$$\psi = \frac{q}{2\pi} \theta = m \theta$$

$$\phi = \frac{q}{2\pi} \ln r = m \ln r$$

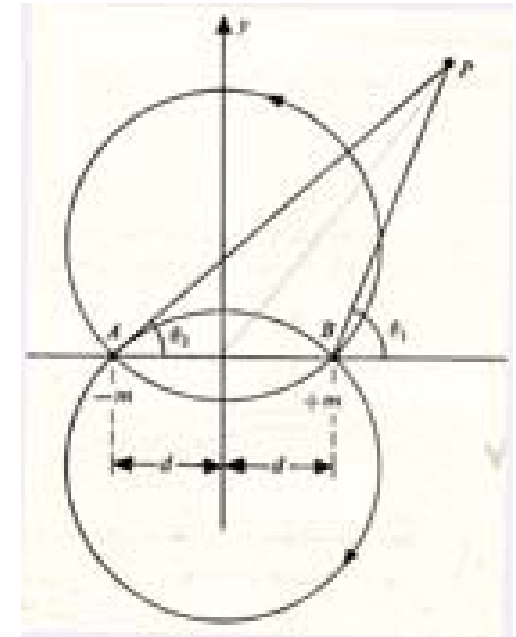
# PLANE - FLOW SOLUTIONS

## a) Source and Sink of equal strength

- Source,  $m$  at  $(-d, 0)$  and a sink,  $m$  at  $(d, 0)$
- For source  $\psi_1 = m\theta_1$ . For sink  $\psi_2 = -m\theta_2$   
(where,  $q =$  strength of source,  $m = q/2\pi$ )
- For combined flow  $\psi = \psi_1 + \psi_2 = m(\theta_1 - \theta_2)$ .

$$\psi = m(\theta_1 - \theta_2)$$

$$\phi = \frac{1}{2} m(\ln r_1 - \ln r_2)$$



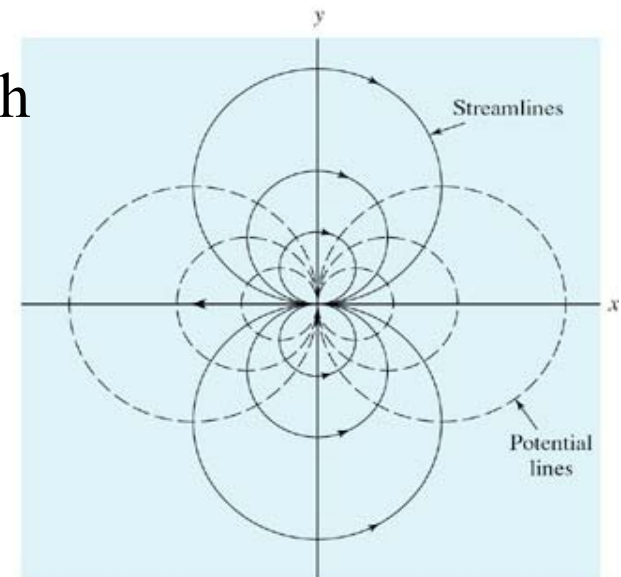
## b) Doublet

It is a special case of source and sink when both approach the origin while keeping the product of strength and distance constant ( $q*d/2\pi = \text{constant} = \mu$ .)

It can be proved that

$$\psi = -\frac{\mu \sin \theta}{r}$$

$$\phi = \frac{\mu}{r} \cos \theta$$



(d) Doublet

# Stream & Potential functions of some standard flows

Types of flow	Stream, $\psi$	Potential $\phi$
Uniform flow in x direction	$U r \sin\theta = U y$	$U r \cos\theta = U x$
Source	$m \theta$	$m \ln r$
Sink	$- m \theta$	$- m \ln r$
Free vortex (anticlockwise)	$-K \ln r$	$K \theta$
Doublet	$-\frac{\mu \sin \theta}{r}$	$\frac{\mu \cos \theta}{r}$

Where  $K = (\Gamma/2\pi)$ ,  $m = q/2\pi$ ; Circulation,  $\Gamma$  can be +ve or -ve.

# Superposition of simple flows

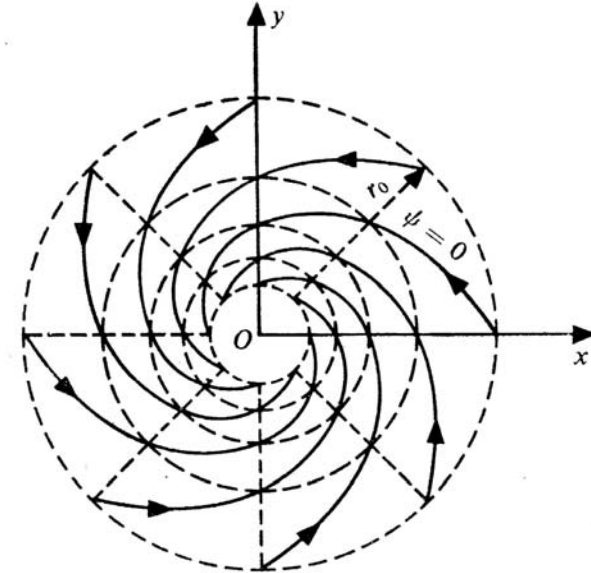
## SPIRAL VORTEX

(Line sink,  $m$ ) + (Free vortex anticlockwise,  $\Gamma$ )

$$\psi = \psi_{sink} + \psi_{vortex} = -m\theta - K \ln r$$

Similarly  $\phi = -m \ln r + K\theta$

Quiz: How would you form an outward spiral?

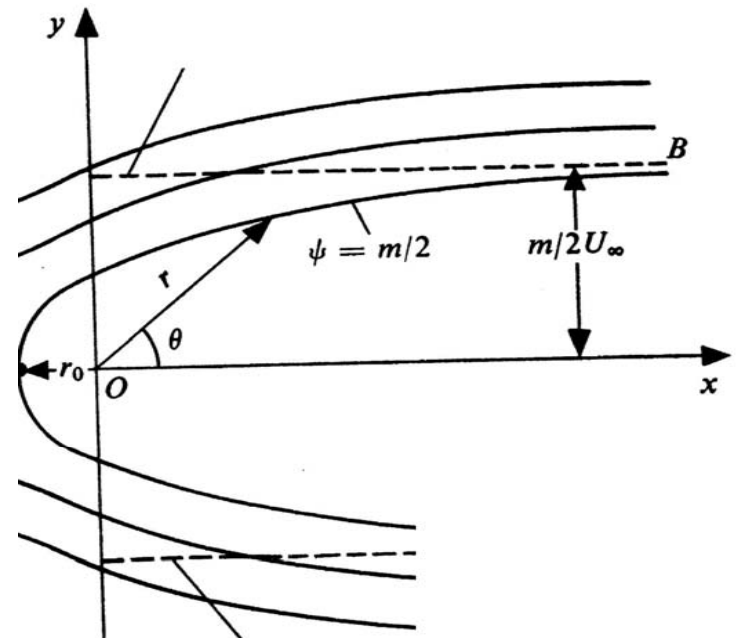


## FLOW OVER HALF BODY

(Uniform flow) + (Source)

$$\Psi = U r \sin\theta + m \theta$$

$$\Phi = U r \cos\theta + m \ln r$$





# Flow past a cylinder

- This can be simulated by superimposing a doublet on a uniform flow

$$\Psi = \Psi_{\text{uniform flow}} + \Psi_{\text{doublet}}$$

$$\Psi = U r \sin \theta - \frac{\lambda \sin \theta}{r} = (U r - \lambda / r) \sin \theta$$

- Streamline on object's surface is zero.** So for the surface  $\sin \theta (U r - \lambda / r) = 0$ ;  
or  $r = (\lambda / U)^{0.5}$ . It is the equation of a circle

- Stagnant points occurs at  $V_{\theta} = 0$ ;

which occurs at 0 & 180.

$$v_{\theta} = -\frac{\partial \Psi}{\partial r} = \left( U + \frac{\lambda}{r^2} \right) \sin \theta$$

Velocity on the cylinder surface is found on substitution

$$v_r = \frac{1}{r} \frac{\partial \Psi}{\partial \theta} = \left( U - \frac{\lambda}{r^2} \right) \cos \theta$$

# Drag

- **Drag** is found by integrating force in flow direction and this integral is found to be zero

$$D = -\int_0^{2\pi} P_s (ba) d\theta \cdot \cos \theta = 0$$

It suggests that there is no drag on a cylinder in cross-flow on a stationery cylinder

- This absurd result is called D'Alembert's Paradox.

# ROTATING CYLINDER

Flow over rotating cylinder = Uniform flow + Free Vortex (anticlockwise) + Doublet

$$\Psi = U r \sin\theta + (-K \ln r) - (\lambda \sin\theta/r)$$

Velocity at a point

$$v_{\theta} = -\frac{\partial \psi}{\partial r} = -U \sin \theta + \frac{K}{r} - \frac{\lambda \sin \theta}{r^2}$$

Vortex flow does not change  $V_r$ . So it remains unchanged.

So, the surface is still defined by  $\lambda = U a^2$

Velocity on its surface is therefore =  $V_{\theta} = -2 U \sin\theta + (K/a)$

Stagnation point occurs where  $V_{\theta} = 0$ ,

$$\sin \theta_s = \frac{K}{2 a U}$$

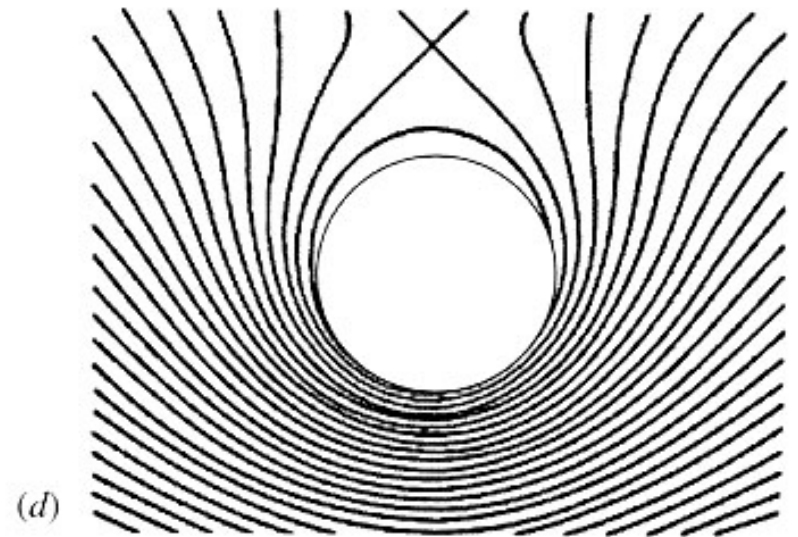
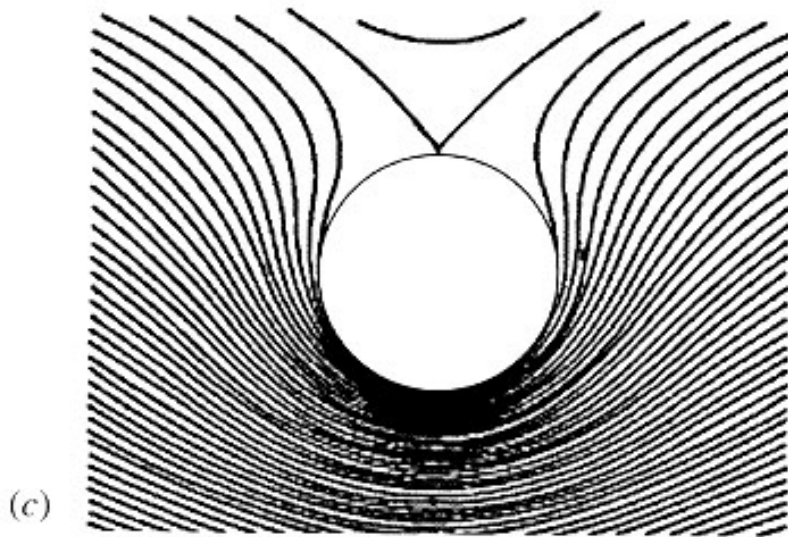
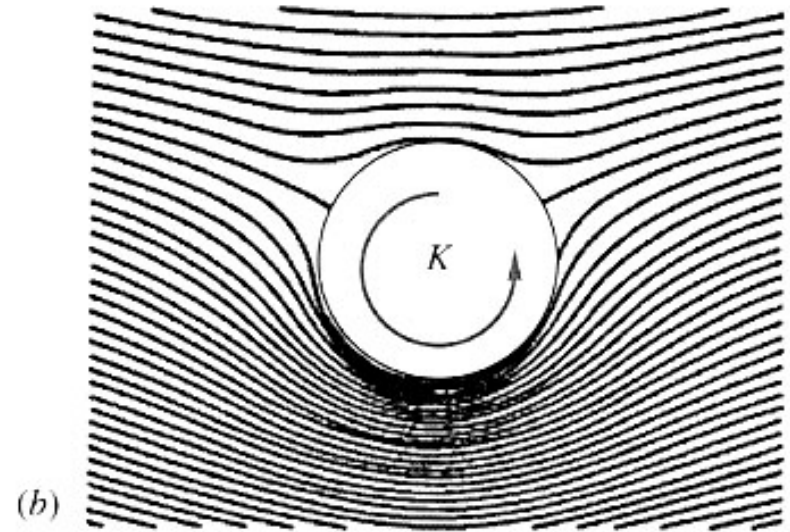
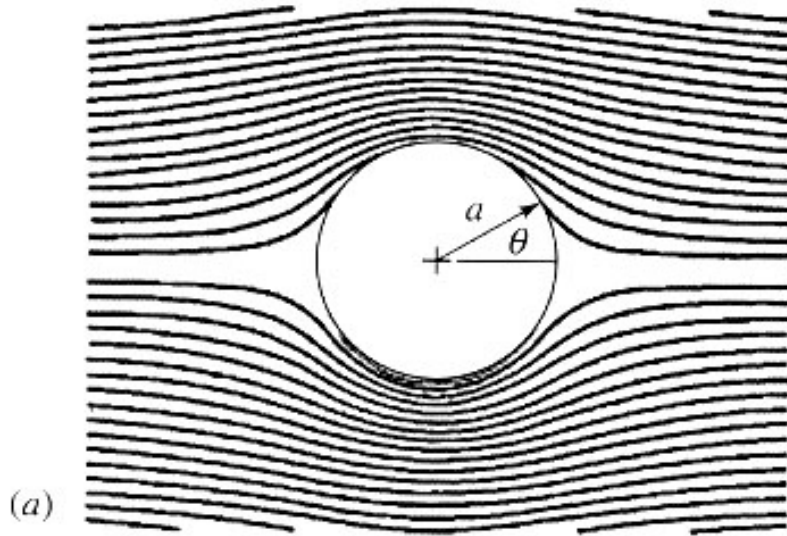
So, no circulation ( $K=0$ ), stagnation occurs at  $\theta_s = 0$  and  $180^\circ$

For rotation

If  $K = aU$ ,  $\sin\theta = 0.5$ ; stagnation occurs at  $\theta_s = 30$  and  $150^\circ$

$$\Gamma = 2 \pi K \text{ where } \Gamma \text{ is circulation}$$

# Stream lines around a rotating cylinder



# ROTATING CYLINDER

Applying Bernoulli's eqn. between the surface (where velocity is  $v_\theta$ ) and freestream we get

$$P_\infty + \frac{1}{2} \rho U^2 = P_s + \frac{1}{2} \rho V_\theta^2 = P_s + \frac{1}{2} \rho \left( -2U \sin \theta + \frac{K}{a} \right)^2$$

$$P_s - P_\infty = \frac{1}{2} \rho U^2 \left( 1 - 4 \sin^2 \theta + 4\beta \sin \theta - \beta^2 \right) \quad \text{where } \beta = \frac{K}{Ua}$$

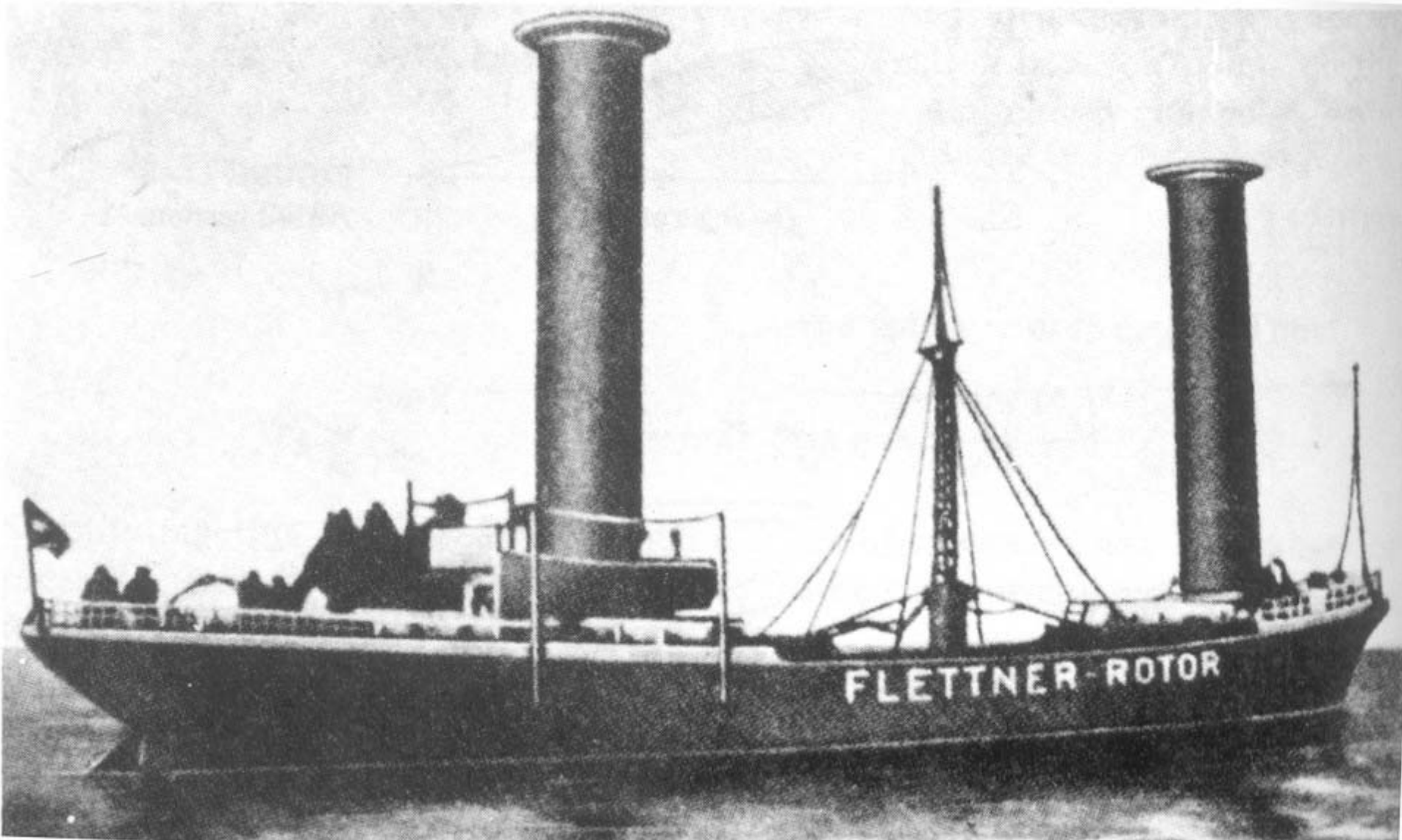
Lift, is found by integrating forces normal to the flow

$$L = - \int_0^{2\pi} (P_s - P_\infty) \sin \theta (ba) d\theta = -\rho U (2\pi K) b = -\rho U \Gamma b$$

$$\frac{L}{b} = -\rho U \Gamma \quad \text{where } a = \text{radius of cylinder} \ \& \ b = \text{length of cylinder}$$

Upward force, Lift on a rotating cylinder is generated when circulation is negative. It is called the **Magnus effect**

Developed by Mr. Flettner in 1927



# Problem - 1

- a) Find the stream function (in polar coordinate) in an uniform flow in  $x$  direction.
  
- b) In a two-dimensional incompressible flow the fluid velocity components are given by  $u=x-4y$  and  $v=-y-4x$ . Find the stream function and show that it is an irrotational flow

## Problem - 2

- In the ideal flow around a half body, the free stream velocity is 0.5 m/s and the strength of the source is  $1/\pi$  m<sup>2</sup>/s. Determine the fluid velocity and its direction at a point  $r = 1$  m and  $\theta = 120$



# Problem on Magnus Effect

- In 1927 a man named Flettner had a ship built with two rotating cylinders to act as sails. The height of the cylinders is 15 m and diameter 2.75 m, The wind speed is 30 km/h, and the speed of the ship is 4 km/h. The cylinders rotate at a speed of 750 rpm by the action of a steam engine below deck. Air density 1.229 kg/m<sup>3</sup>.
- Find the maximum possible propulsive thrust on the ship from the cylinders.
  - [Shames p-597]

# Home work (White 4.72)

- A coastal power plant takes in cooling water through a vertical perforated manifold. The total volume flow intake is  $110 \text{ m}^3/\text{s}$ . Currents of  $0.25 \text{ m/s}$  flow past the manifold. Estimate how far downstream and how far normal to the paper the effects of the intake are felt in the ambient  $8 \text{ m}$  deep waters.  
[8.75 m, 55 m]